

NASA Technical Memorandum 102180

1N-18

7291

P13

A New Method for Determining Which Stars Are Near a Star Sensor Field-of-View

Russell E. Yates, Jr.
John D. Vedder

February 1991



(NASA-TM-102180) A NEW METHOD FOR
DETERMINING WHICH STARS ARE NEAR A STAR
SENSOR FIELD-OF-VIEW (NASA) 13 p CSCL 22B

N91-20201

Unclas
G3/18 0007291

NASA Technical Memorandum

A New Method for Determining
Which Stars Are Near a Star
Sensor Field-of-View

Russell E. Yates, Jr.
Lyndon B. Johnson Space Center
Houston, Texas

John D. Vedder
McDonnell Douglas Space Systems Company
Houston, Texas

National Aeronautics and Space Administration
Lyndon B. Johnson Space Center
Houston, Texas

February 1991

CONTENTS

Section		Page
	Abstract.....	1
1.0	Introduction.....	2
2.0	Background.....	2
3.0	Method.....	3
4.0	Algorithm.....	4
5.0	Test Results.....	6
6.0	Conclusions.....	7
	Bibliography.....	7

TABLES

Table		Page
1	Comparison of the scalar-product method and the new method using a math coprocessor.....	6
2	Comparison of the scalar-product method and the new method without using a math coprocessor.....	6

FIGURES

Figure		Page
1	Intersection of orthogonal bands containing star sensor FOV.....	3
2	Star sensor FOV within coordinate interval $[x_{min}, x_{max}]$	4

ABSTRACT

A new method is described for determining which stars in a navigation star catalog are near a star sensor field-of-view (FOV). This method assumes that an estimate of spacecraft inertial attitude is known. Vector component ranges for the star sensor FOV are computed, so that stars whose vector components lie within these ranges are near the star sensor FOV. This method requires no presorting of the navigation star catalog, and is more efficient than traditional methods.

1.0 INTRODUCTION

Many spacecraft maintain inertial attitude knowledge by sighting navigation stars with star sensors. The process of determining exactly which navigation stars are in a star sensor field-of-view (FOV), given an estimate of spacecraft attitude, is computer intensive. Consequently, spacecraft first determine which navigation stars are *near* the star sensor FOV, and then determine which nearby stars are *in* the star sensor FOV. This paper presents a new method of determining which navigation stars are near a star sensor FOV that is more efficient than previous methods.

2.0 BACKGROUND

There are two classes of methods for determining which navigation stars are near a star sensor FOV: those methods that *do not* require presorting the navigation star catalog and those methods that *do* require presorting the navigation star catalog.

The primary no-presorting method is the scalar-product method. This method is used on the Space Shuttle. It takes advantage of the simple scalar-product relation

$$\theta = \cos^{-1}(\mathbf{b} \cdot \mathbf{s})$$

where \mathbf{b} is a unit star sensor boresight vector in inertial coordinates, \mathbf{s} is a unit navigation star position vector in inertial coordinates, and θ is the angle between \mathbf{b} and \mathbf{s} . If θ is less than the angular radius of the star sensor FOV, then \mathbf{s} is near the star sensor FOV. Obviously, the scalar-product method is simple, but it is also computer intensive.

There are numerous presorting methods that determine which navigation stars are near a star sensor FOV. Although these methods are normally less computer intensive than the scalar-product method, they do introduce some additional problems. For example, some algorithms require that the navigation star catalog be presorted into predefined zones on the celestial sphere. These algorithms are complicated and require additional computer data storage to index the navigation stars.

It is desirable to have a method for determining which navigation stars are near a star sensor FOV that is simple, fast, and requires no presorting of the navigation star catalog. The method described below satisfies these requirements: it is a no-presorting method that is faster than the scalar-product method.

3.0 METHOD

Let us assume that we have a star sensor with a known FOV and an associated navigation star catalog. In addition, let x , y , and z be the principal axes of the inertial coordinate system in which the coordinates of the navigation stars are cataloged. Then we may define orthogonal bands on the celestial sphere that are normal to the x , y , and z axes; and whose unique intersection Σ bounds a circular cap on the celestial sphere that is circumscribed about the star sensor FOV. This concept is illustrated in Figure 1 below.

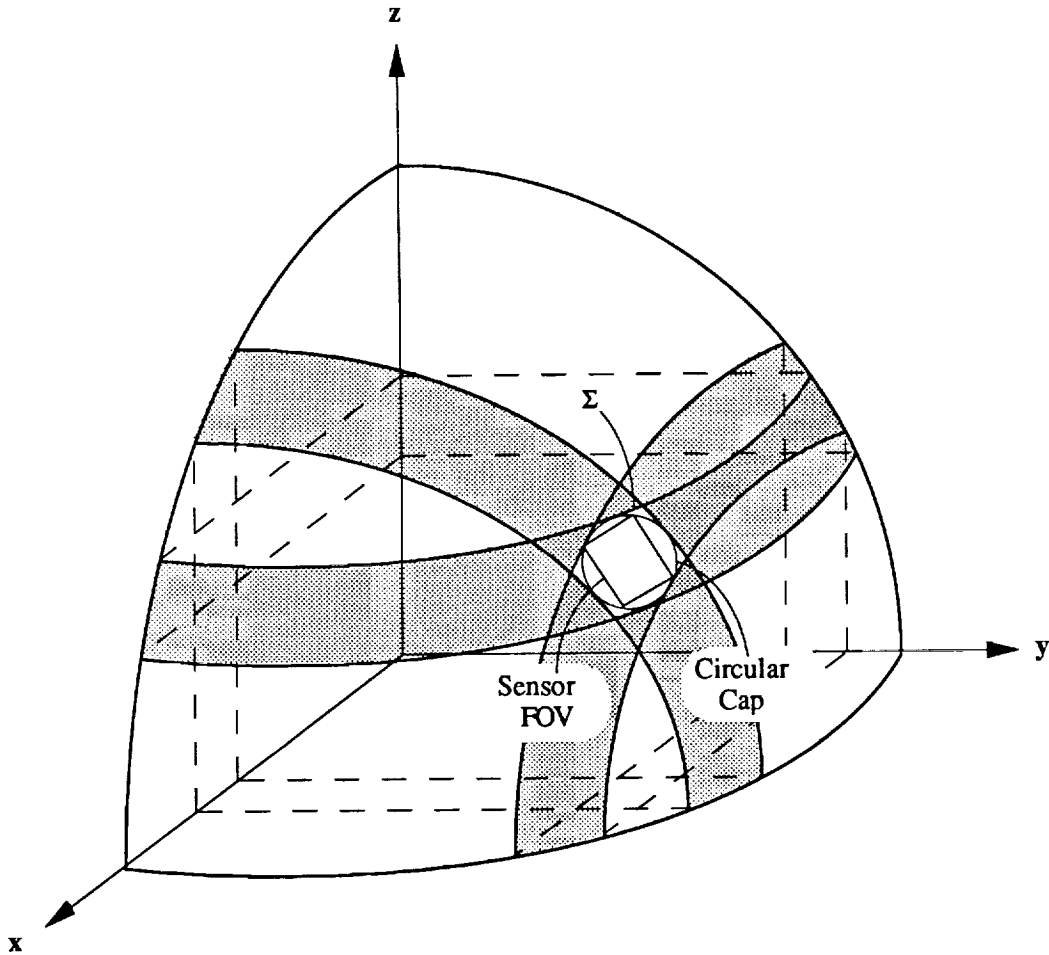


Figure 1 - Intersection of orthogonal bands containing star sensor FOV.

To determine which navigation stars are near the star sensor FOV, we need simply determine which navigation stars are in Σ . Since the orthogonal bands illustrated above are normal to the x , y , and z axes, Σ may be defined by its projections onto x , y , and z . Let the intervals $[x_{\min}, x_{\max}]$, $[y_{\min}, y_{\max}]$, and $[z_{\min}, z_{\max}]$ be the projections of Σ onto x , y , and z respectively. Then to determine which navigation stars are in Σ , and thus near the star sensor FOV,

we need simply search the navigation star catalog for stars whose inertial coordinates are contained in $[x_{\min}, x_{\max}]$, $[y_{\min}, y_{\max}]$, and $[z_{\min}, z_{\max}]$.

4.0 ALGORITHM

To compute the coordinate interval $[x_{\min}, x_{\max}]$, let $\mathbf{b} = (b_x, b_y, b_z)$ be the star sensor boresight vector in inertial coordinates, α_x be the angle between the boresight vector and the x-axis, and β be the diagonal radius of the star sensor FOV. These parameters are illustrated in Figure 2 below.

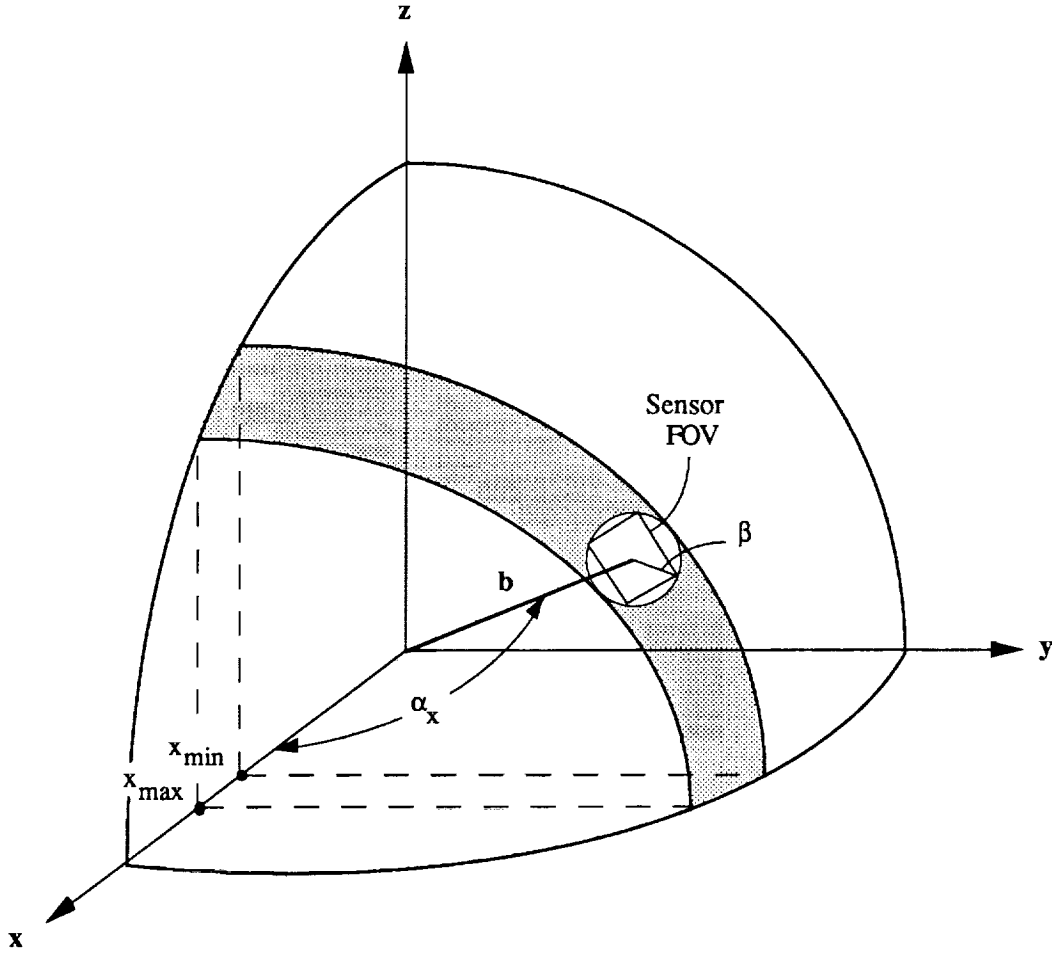


Figure 2 - Star sensor FOV within coordinate interval $[x_{\min}, x_{\max}]$.

The angle between the boresight vector and the x-axis, α_x , is given by

$$\alpha_x = \cos^{-1}(b_x), \quad \text{where } 0 \leq \alpha_x \leq \pi.$$

The coordinate interval $[x_{\min}, x_{\max}]$ may be computed as follows:

```

if  $(\alpha_x - \beta) < 0$  then
     $x_{\max} = 1$ 
     $x_{\min} = \cos(\alpha_x + \beta)$ 
elseif  $(\alpha_x + \beta) > \pi$  then
     $x_{\max} = \cos(\alpha_x - \beta)$ 
     $x_{\min} = -1$ 
else
     $x_{\max} = \cos(\alpha_x - \beta)$ 
     $x_{\min} = \cos(\alpha_x + \beta)$ 
endif

```

The intervals $[y_{\min}, y_{\max}]$ and $[z_{\min}, z_{\max}]$ can be computed in a completely analogous manner.

Once the coordinate intervals $[x_{\min}, x_{\max}]$, $[y_{\min}, y_{\max}]$, and $[z_{\min}, z_{\max}]$ have been computed, we must search through the navigation star catalog for stars whose inertial coordinates are contained in these intervals, and thus in Σ and near the star sensor FOV. The following is an example of how this process may be implemented.

```

loop Star Catalog
    if  $s_x > x_{\min}$  then
        if  $s_x < x_{\max}$  then
            if  $s_y > y_{\min}$  then
                if  $s_y < y_{\max}$  then
                    if  $s_z > z_{\min}$  then
                        if  $s_z < z_{\max}$  then
                            Star is near star sensor FOV
                        endif
                    endif
                endif
            endif
        endif
    endif
end loop

```

The algorithm above may be improved by ordering the "if" statements so that the least-probable "if" statements appear first.

5.0 TEST RESULTS

The results of a comparison between the scalar-product method and the new method described above are given in Table 1 and Table 2 below. The tests were run on a Macintosh IIfx using Absoft MacFortran/020, Version 2.4. A constant star sensor FOV diagonal radius of 8 degrees was assumed. Table 1 summarizes data generated using a math coprocessor, and Table 2 summarizes data generated without using a math coprocessor.

Table 1. Comparison of the scalar-product method and the new method using a math coprocessor.

Number of Trials	Number of Stars	Scalar-Product Method (sec)	New Method (sec)	Ratio of Times (S-P : New)
1000	100	4.72	1.93	2.4 : 1
1000	1000	46.85	16.33	2.9 : 1
1000	5000	234.18	80.05	2.9 : 1

Table 2. Comparison of the scalar-product method and the new method without using a math coprocessor.

Number of Trials	Number of Stars	Scalar-Product Method (sec)	New Method (sec)	Ratio of Times (S-P : New)
1000	100	27.33	14.37	1.9 : 1
1000	1000	273.17	26.17	10.4 : 1
1000	5000	1365.85	78.73	17.3 : 1

The test results indicate the new method is superior to the scalar-product method. Improvement was shown over the scalar-product method, both with and without a math coprocessor.

6.0 CONCLUSIONS

A new method for quickly identifying which stars in a navigation star catalog are near a star sensor FOV has just been described. This method is faster than the traditional scalar-product method, yet retains the advantage of requiring no presorting of the navigation star catalog.

If a faster algorithm is desired, the navigation star catalog may be presorted by unit vector components. This will increase computational speed, but it is more complicated and will require additional computer data storage to index the stars.

The method is presented as an application for onboard spacecraft attitude determination, but it may also apply to other areas such as astronomy.

BIBLIOGRAPHY

"Space Shuttle Orbiter Operational Level C Functional Subsystem Software Requirements; Guidance, Navigation, and Control; Part E, Star Tracker Subsystem Operating Program." NASA STS 83-0014C, Vol. 2, June 1990.

Wertz, James R. Spacecraft Attitude Determination and Control. Dordrecht, Holland: D. Reidel, 1978.



National Aeronautics and
Space Administration

REPORT DOCUMENTATION PAGE

1. Report No. TM 102180		2. Government Accession No.		3. Recipient's Catalog No.	
4. Title and Subtitle A New Method for Determining Which Stars are Near a Star Sensor Field-of-View				5. Report Date February 1991	
				6. Performing Organization Code EG4	
7. Author(s) Russell E. Yates, Jr. John D. Vedder*				8. Performing Organization Report No. S-631	
9. Performing Organization Name and Address Lyndon B. Johnson Space Center Navigation Control and Aeronautics Division Houston, Tx 77058				10. Work Unit No.	
				11. Contract or Grant No.	
12. Sponsoring Agency Name and Address National Aeronautics and Space Administration Washington, D.C. 20546				13. Type of Report and Period Covered Technical Memorandum	
				14. Sponsoring Agency Code	
15. Supplementary Notes *McDonnell Douglas Space Systems Company Houston, TX					
16. Abstract A new method is described for determining which stars in a navigation star catalog are near a star sensor field-of-view (FOV). This method assumes that an estimate of spacecraft inertial attitude is known. Vector component ranges for the star sensor FOV are computed, so that stars whose vector components lie within these ranges are near the star sensor FOV. This method requires no presorting of the navigation star catalog, and is more efficient than traditional methods.					
17. Key Words (Suggested by Author(s)) attitude determination, star sensor, star tracker, star catalog, data sorting				18. Distribution Statement Unclassified - unlimited Subject Category 18	
19. Security Classification (of this report) Unclassified		20. Security Classification (of this page) Unclassified		21. No. of pages	22. Price

For sale by the National Technical Information Service, Springfield, VA 22161-2171

